

Michael Kröning Material Degradation of Nuclear Structures - Mitigation by Nondestructive Evaluation TPU Lecture Course 2014



MATERIAL CHARACTERIZATION

4.	Mitigation Strategies – The world is never perfect
4.1.	Structure Design and NDT
4.2.	Application of NDT
4.3.	Limits of NDT
4.4.	Quantitative NDT
4.5.	Material Characterization
4.6.	Case Study – Inspection by Cause



Material Testing and Inspection

Destructive Testing

Defines Material Utilization

Provides Insight to Microstructure

It relies on representative samples

Material Specification

Material and System Tests

Material Development

Nondestructive Testing

Defines Quality Standards

Flaw Inspection (QNDT

Full Volume Inspection



Quality Control

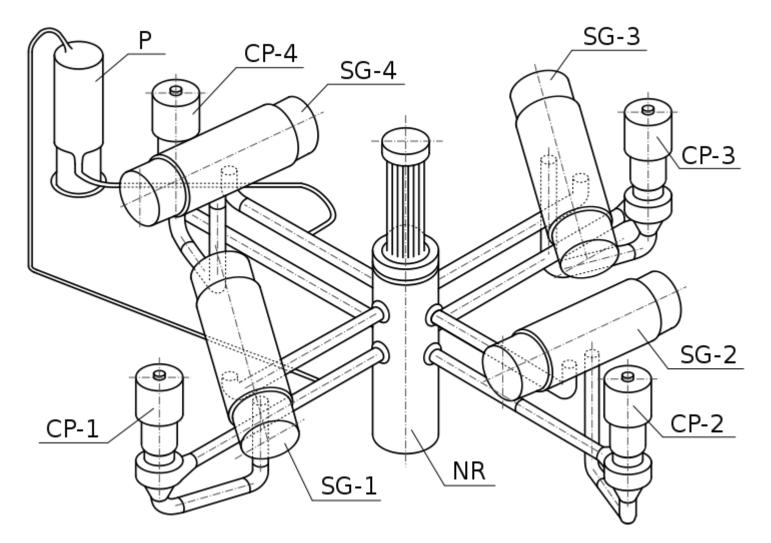
Maintenance

Life-Time Extension

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4 Loop Pressurized Water Reactor (VVER 1000)



VVER-1000-Stereometric.svg

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Central Ring with Outlet and Inlet Nozzles



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Closure Head with Control Rod and Instrumentation Nozzles



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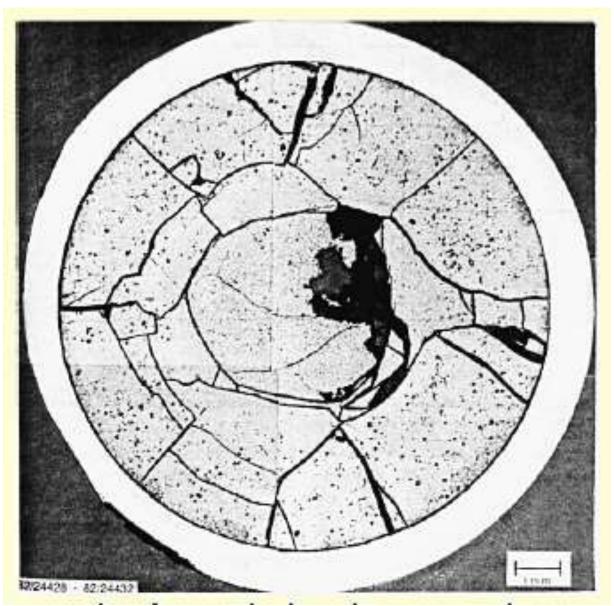
Core Plate



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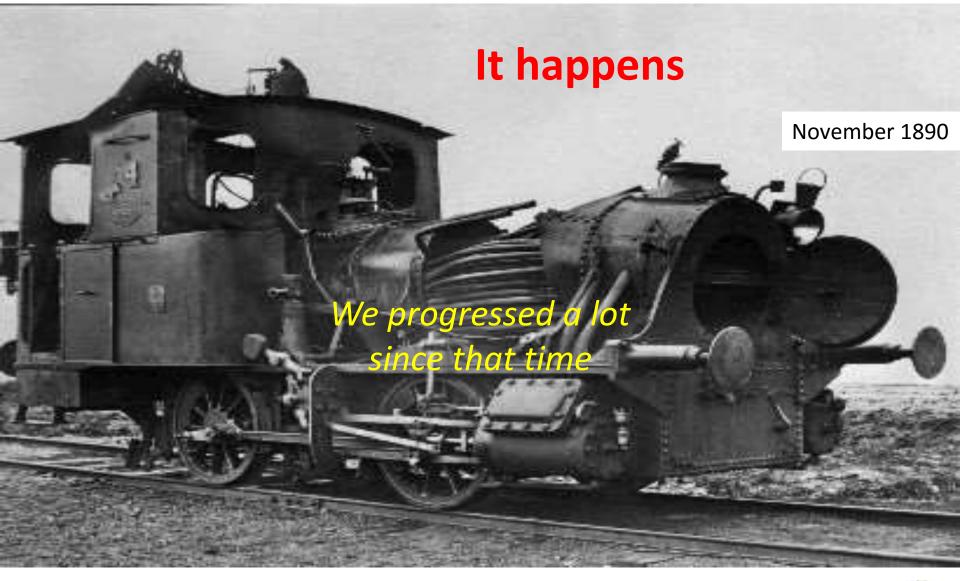
Fuel Cross Section



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Mitigation



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CORE LAVA



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Structural Integrity

Structural integrity is the ability of a structure or a component to withstand a designed service load, resisting structural failure



Structural integrity is a concept often used in engineering, to produce items that will function adequately for their

designed purposes

for a

desired service life.



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System Engineering Rules for Reliable and Safe Structures

Reliability

theoretically defined as the probability of failure, the frequency of failures, or in terms of availability, a probability derived from reliability and maintainability. Reliability plays a key role in cost-effectiveness of systems.

Safety

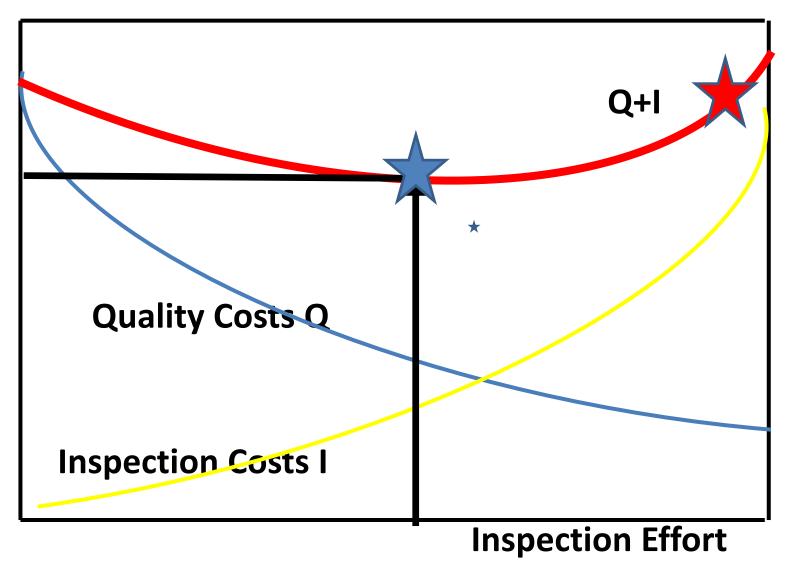
relates to only very specific system hazards that could potentially lead to severe accidents. It deals with dangerous events for life and environment.

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Costs

Structure Design & NDT

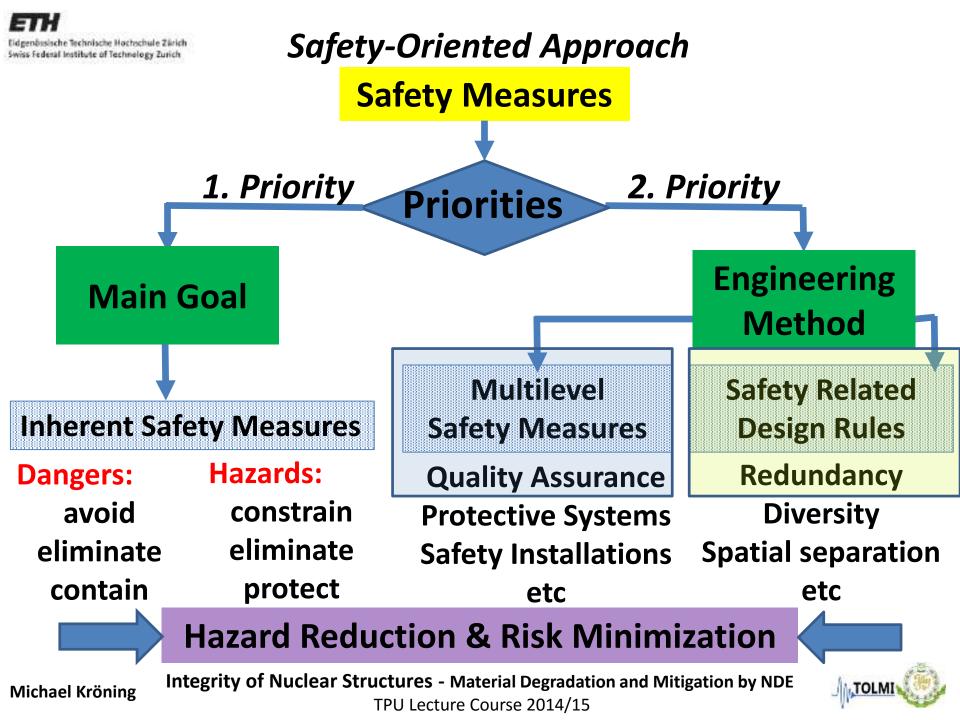


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SAFETY ENGINEERING

DESIGN

- Strictly controlled
- Common cause failure
- Passive systems
- Inherent safety

OPERATION

- Maintenance
- Degradation control
- Unexpected events

MANUFACTURING

- Regulations and codes
- Quality management
- Quality certification
- Independent assessment

DISPOSAL

- Dismantling
- Renaturization
- Waste management



Structure Design & NDT Types of Failures

Overload of Structure (critical stress level)

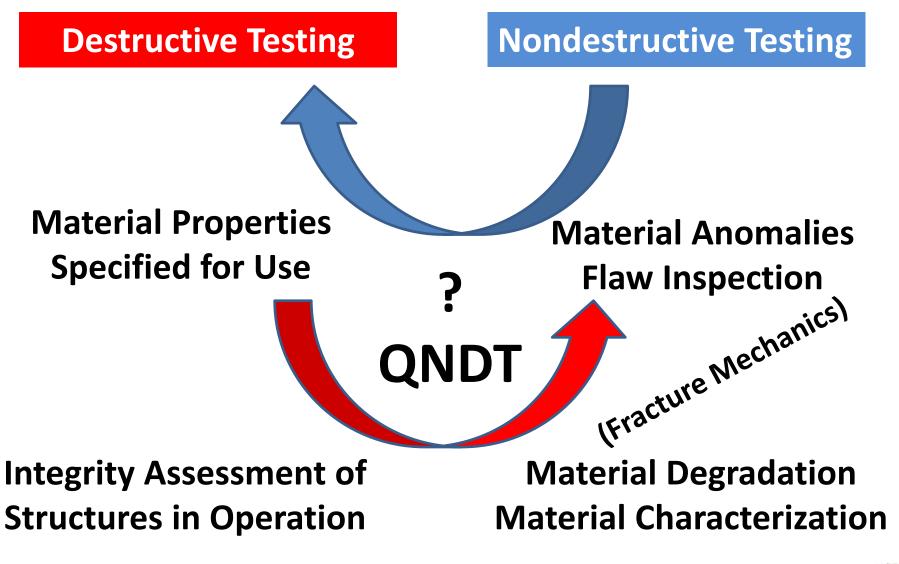
- Structure Instability (due to design or material choice) causing the structure to fail from fatigue or corrosion
- Manufacturing Errors: improper selection of materials, incorrect sizing, improper heat treating, failing to adhere to the design, or shoddy workmanship
- Defective Materials
- Unexpected Problems

(Vandalism, sabotage, and natural disasters)

HOW TO DETECT CAUSES or THE ROLE OF NDT

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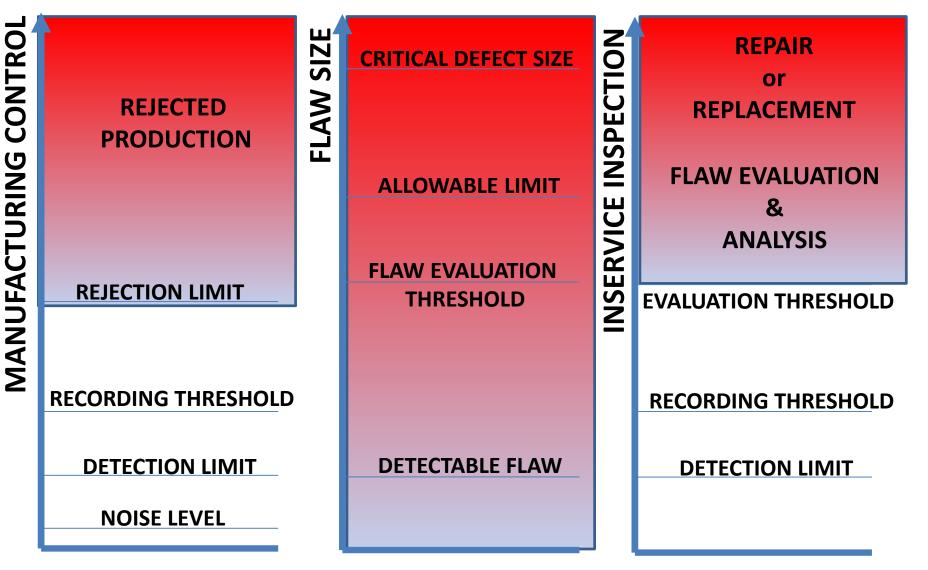
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INDICATION-FLAW-DEFECT



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CONTRIBUTION OF NDT

Crack growth, as shown by fracture mechanics, is exponential in nature (Paris' law).

A desire for reasonable inspection intervals, combined with the exponential growth of cracks in structure has led to the development of non-destructive testing methods which allow inspectors to look for very tiny cracks.

Examples of this technology include eddy current, ultrasonic, dye penetrant, and X-ray inspections.

HOWEVER:

CAN WE CHARACTERIZE MATERIAL DEGRADATION BY MEASURING MICROSTRUCTURE DEPENDANT *AK*?

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CONTRIBUTION OF NDT

STILL LIMITED

FLAW TYPE MATERIAL STRUCTURE ACCESS FLAW INITIATION & GROWTH MATERIAL DEGRADATION CAUSE EVALUATION

TARGETED

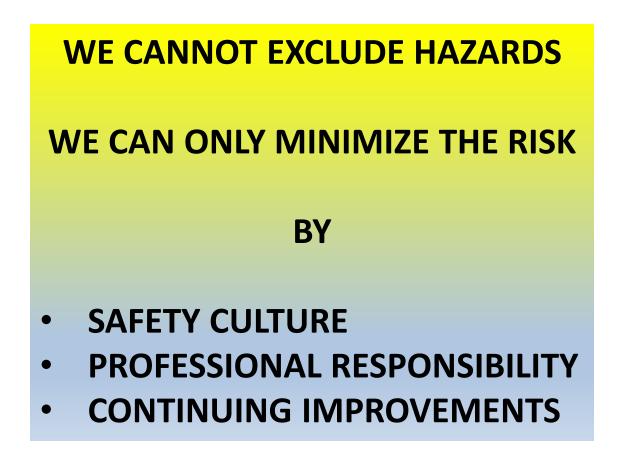
CODES & REQUIREMENTS CERTIFICATION & VALIDATION DETECTION & EVALUATION

QUANTITATIVE NDT NEW METHODS HEALTH MONITORING

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CONCLUSION I



Мтоги

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CONCLUSION II

WE MUST ASSURE THE QUALITY WE MUST CONTROL THE STRUCTURAL STATE

BY

NONDESTRUCTIVE FLAW EVALUATION
NONDESTRUCTIVE MATERIAL CHARACTERIZATION

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CONCLUSION III

CAN WE EVALUATE FLAWS QUANTITATIVELY? CAN WE CHARACTERIZE MATERIAL PROPERTIES?

THERE ARE MANY CHALLENGES FOR YOUNG SCIENTISTS & PIONEERS

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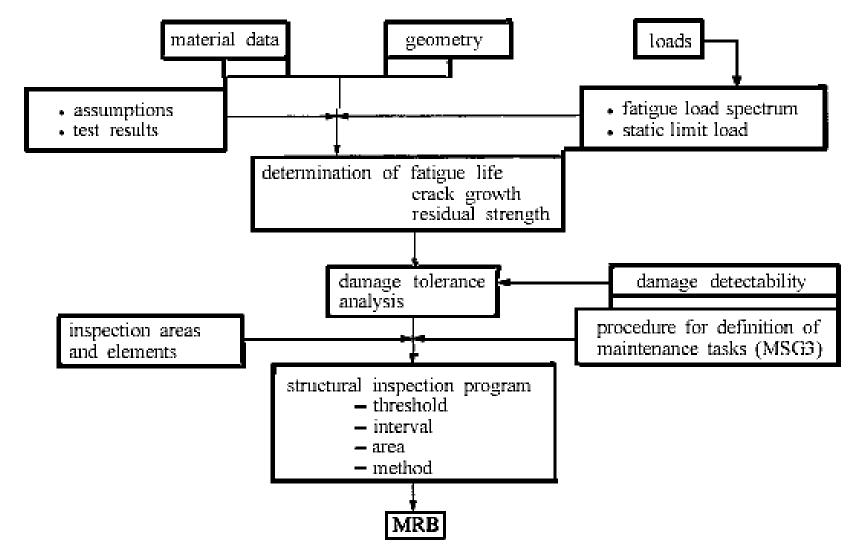
2.	Degradation of nuclear structures during operation
2.1.	Aging, Neutron Embrittlement, Structural Material Parameters
2.2.	Stress Corrosion Cracking
2.3.	Fatigue
2.4.	(Unexpected events)



2.	Degradation of airplane structures during operation
2.1.	Fatigue Life (Endurance)
2.2.	Damage Tolerance Capability
2.3.	Corrosion Resistance
2.4.	(Unexpected events)



Development of Structure Inspection Program



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AGING AIRCRAFTS

DESIGN PRINCIPLE SPECIAL ISSUES

• Safe Life

Damage Tolerant

• Fail Safe





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AGING AIRCRAFTS

Fail Safe Design

Fail safe design is an integral part of any design process. It ensures that the system being designed is both safe and reliable for the user, as well as being economically and environmentally sound.

Fail safe design is essentially the process we used to eliminate potential hazards to the system from the most common scenarios the system is likely to be exposed to in the location it will be in. Furthermore, design of the system had to account for the human factor using technology safely and effectively.

Fail safe design is sometimes a moral and/or ethical dilemma. Fail safe design can sometimes make a project cost more and some members may want to do away with expensive fail safe precautions. This can lead to exposure to risks and possibilities that could have been prevented.

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Aviation Glossary

Fail-Safe Design Concept

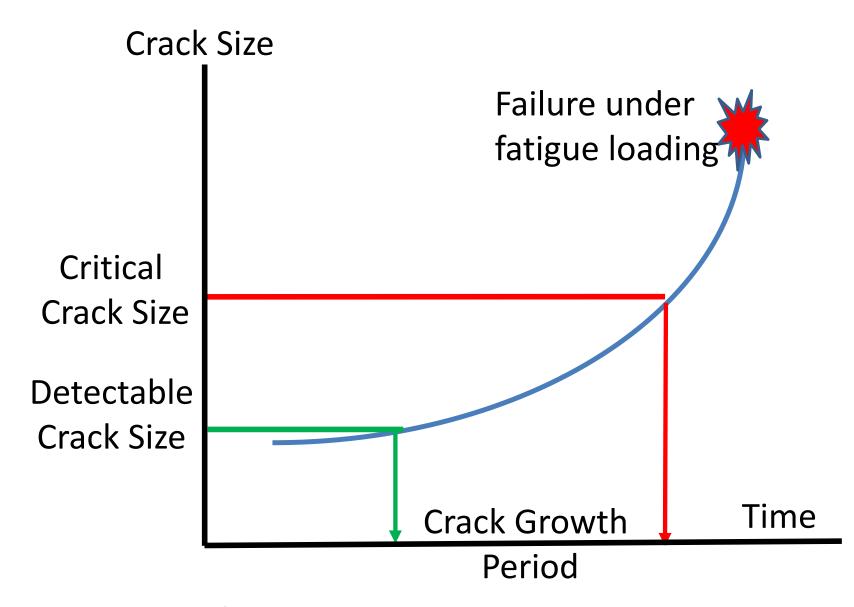
Fail-Safe Design Concept: Fundamental to the notion of safety-critical systems in certification is the fail-safe design concept, which "considers the effects of failures and combinations of failures in defining a safe design." The concept has a different meaning for structures than for systems: fail-safe for structures is concerned with residual strength after sustaining damage; fail-safe for systems is concerned with the functional implications of a failure condition and its probability of occurrence. Although both concepts have the same goal— a safe design—the approaches to achieving that goal are different.

Fail-safe for structures is governed by 14 CFR 25.571 and the methods of compliance are outlined in AC 25.571-1C. In general, the structural components of an airplane (such as the airframe and wings) are designed such that "an evaluation of the strength, detail design, and fabrication must show that catastrophic failure due to fatigue, corrosion, manufacturing defects, or accidental damage, will be avoided throughout the operational life of the airplane." However, after the 1988 Aloha Airlines flight 243 accident, where 18 feet of the upper crown skin and structure separated from the fuselage, there has been a greater emphasis on damage tolerance. A damage tolerance evaluation of structure ensures that "should serious fatigue, corrosion, or accidental damage occur within the design service goal of the airplane, the remaining structure can withstand reasonable loads without failure or excessive structural deformation until the damage is detected."



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Principle of damage tolerance investigation



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Damage Tolerance Requirements



- Structural design according to fatigue and damage tolerance requirements
- Evaluation of the structure by analysis supported tests
- Definition of a structural inspection program



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Damage Tolerance Evaluation

"The ultimate purpose of the damage tolerance evaluation Is the development of a

recommended structural inspection program

considering

probable damage locations, crack initiation mechanisms, crack growth time histories, and crack detectability"

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Regulations and Recommendations

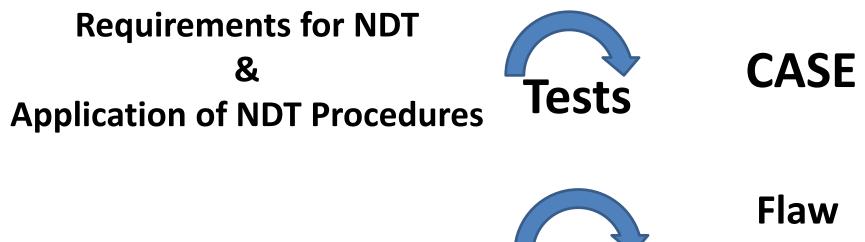
The (internationally) harmonized regulation and advisory circular require:

'An evaluation of the strength, detailed design, and fabrication must show that a catastrophic failure due to fatigue, corrosion, or accidental damage, will be avoided throughout the operational life of the airplane.'

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Conclusion



State of Usage



Detection QNDT

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Literature

H.J. Schmidt, B. Schmidt-Brandecker, G. Tober: *Design of Modern Aircraft Structure and the Role of NDI*, NDT.net, vol. 4 No. 6, 1999

